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C =====
C
C      ABAQUS UMAT for Linear Elastic Isotropic Material
C
C      Code adapted from:
C          http://birch.seas.harvard.edu/files/Writing%20a%20UMAT.pdf
C
C      Subroutine written by:
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C
C =====
C
C      SUBROUTINE UMAT(STRESS, STATEV, DDSDDE, SSE, SPD, SCD, RPL,
C      1 DDSDDT, DRPLDE, DRPLDT, STRAN, DSTRAN, TIME, DTIME, TEMP, DTEMP,
C      2 PREDEF, DPRED, CMNAME, NDI, NSHR, NTENS, NSTATV, PROPS, NPROPS,
C      3 COORDS, DROT, PNEWDT, CELENT, DFGRD0, DFGRD1, NOEL, NPT, LAYER,
C      4 KSPT, KSTEP, KINC)
C          INCLUDE 'ABA_PARAM.INC'
C          CHARACTER*80 CMNAME
C          DIMENSION STRESS(NTENS), STATEV(NSTATV), DDSDDE(NTENS, NTENS),
C      1 DDSDDT(NTENS), DRPLDE(NTENS), STRAN(NTENS), DSTRAN(NTENS),
C      2 PREDEF(1), DPRED(1), PROPS(NPROPS), COORDS(3), DROT(3, 3),
C      3 DFGRD0(3, 3), DFGRD1(3, 3), EELAS(6), EPLAS(6), FLOW(6)
C
C =====
C      Elastic constants for User Material
C      Young Modulus
C          EMOD=PROPS(1)
C      Poisson Ratio
C          ENU=PROPS(2)
C
C =====
C      ELASTIC STIFFNESS MATRIX (in terms of E, v)
C
C [E(1-v)]/[(1+v)(1-2v)] [Ev]/[(1+v)(1-2v)] [Ev]/[(1+v)(1-2v)] 0 0 0
C [Ev]/[(1+v)(1-2v)] E(1-v)/[(1+v)(1-2v)] [Ev]/[(1+v)(1-2v)] 0 0 0
C [Ev]/[(1+v)(1-2v)] [Ev]/[(1+v)(1-2v)] [E(1-v)]/[(1+v)(1-2v)] 0 0 0
C 0 0 0 [E]/[2(1+v)] 0 0
C 0 0 0 0 [E]/[2(1+v)] 0
C 0 0 0 0 0 [E]/[2(1+v)]
C
C ----- Hooke's Law in Stiffness Form:
C      http://www.efunda.com/formulae/solid_mechanics/mat_mechanics/hooke_isotropic.cfm
C
C      DO K1=1,NDI
C          DO K2=1,NDI
C              DDSDDE(K2,K1)=(EMOD*(ENU))/((1+ENU)*(1-2*ENU))
C          END DO
C          DDSDDE(K1,K1)=(EMOD*(1-ENU))/((1+ENU)*(1-2*ENU))
C      END DO
C      DO K1=NDI+1,NTENS
C          DDSDDE(K1,K1)=(EMOD)/(2*(1+ENU))
C      END DO
C
C =====
C      STRESS CALCULATION
C
C [Sig11] [ E(1-v)/(1+v)(1-2v) Ev/(1+v)(1-2v) Ev/(1+v)(1-2v) 0 0 0 ] [eps11]
C [Sig22] [ Ev/(1+v)(1-2v) E(1-v)/(1+v)(1-2v) Ev/(1+v)(1-2v) 0 0 0 ] [eps22]
C [Sig33] = [ Ev/(1+v)(1-2v) Ev/(1+v)(1-2v) E(1-v)/(1+v)(1-2v) 0 0 0 ] [eps33]
C [Tau12] [ 0 0 0 E/2(1+v) 0 0 ] [gam12]
C [Tau13] [ 0 0 0 0 E/2(1+v) 0 ] [gam13]
C [Tau23] [ 0 0 0 0 0 E/2(1+v) ] [gam23]
C
C ----- DO K1=1,NTENS
C         DO K2=1,NTENS
C             STRESS(K2)=STRESS(K2)+DDSDDE(K2,K1)*DSTRAN(K1)
C         END DO
C     END DO
C
C =====
C      RETURN
C

```